Engineered Polymer Composites Through Electrospun Nanofiber Coating of Fiber Tows

NASA Aeronautics Research Mission Directorate (ARMD)

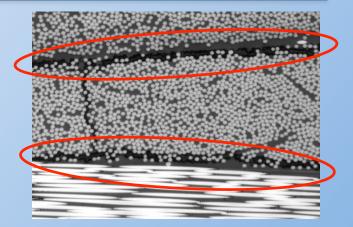
FY12 Seedling Phase I Technical Seminar

July 9-11, 2013

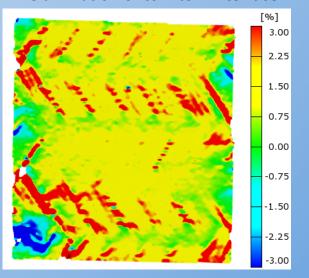


Motivation/Impact

- Failure of advanced composite structures is often dominated by interfacial failure such as delamination.
- Local features such as holes, notches, and defects can cause failure of the composite at loads well below the fiber strength.
- Toughening of the material locally at the interface could dramatically improve the mechanical performance of the material without bulk modification of the matrix.
- This could lead to a relatively inexpensive way to improve performance without adversely affecting processing.
- Improved mechanical durability and strength would directly result in lower weight structures and improved efficiency in many applications such as engine fan blades.



Delamination of tow-tow interface



Local edge damage in a PMC



Innovation and Technical Approach

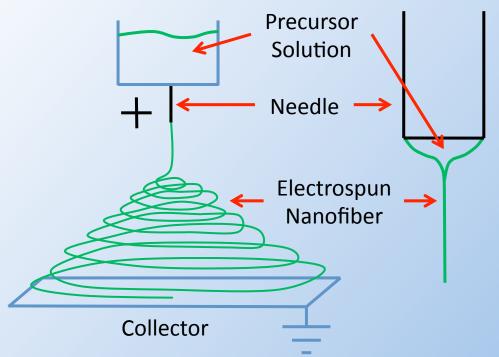
- Develop a method for directly depositing thermoplastic nanofiber on continuous fiber materials for composite interface toughening.
- Build a machine capable of producing larger quantities of the coated material (1000's of feet).
- Coat enough material to produce filament wound tubes for mechanical tests.
- Investigate possible interface toughening capability.

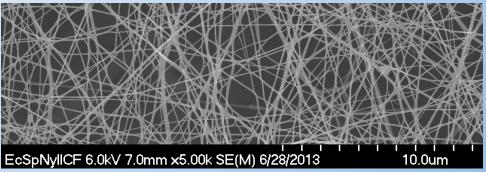




Electrospinning Process

NASA Aeronautics Research Institute





Factors in Electrospinning

- Solution viscosity
- Liquid solvent vapor pressure
- Ambient solvent vapor concentration
- Applied voltage
- Temperature
- Collector distance
- Polymer mechanical properties
- Solution conductivity
- Ambient gas flow conditions



Early Work

NASA Aeronautics Research Institute

- GRC Fast Track demonstrated initial feasibility.
- ARMD Seedling provided opportunity to scale up to usable material quantities.

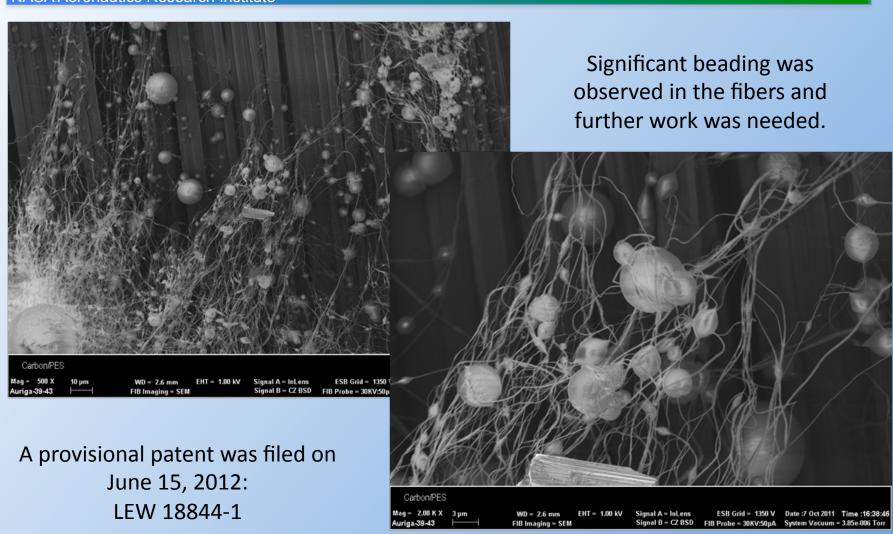
Polyethersulfone and DMF solution was electrospun using a ring of 8 needles onto carbon fiber







Early Results





Material Selection

- Towpreg consisting of 12k T700s carbon fiber with UF3325 resin was purchased from TCR Composites.
- The material has a 12 month shelf life at room temperature and can be cured as low as 132
 Celsius. A cure at 143 Celsius for 2 hours was used.
- Nylon 11 was selected as the nanofiber toughener because of favorable mechanical properties, high melt temperature, and solvent resistance.
- A mixture of 3 parts dichloromethane and 1 part formic acid by weight was used to dissolve the Nylon 11.
- Different solutions were tried and a 2.5 % solution by weight was selected.

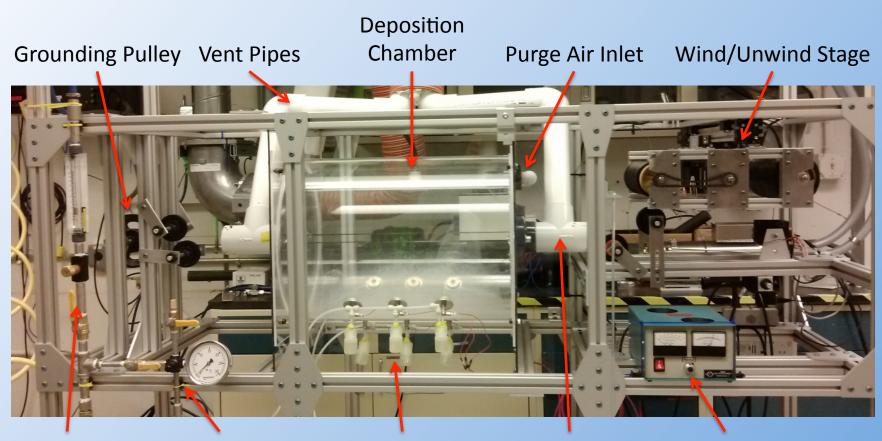


Fan Blade Leading Edge Impact Test Coupon



Electrospinning Machine

NASA Aeronautics Research Institute



Purge Air Control

Control

Solution Feed Needles w/ Attached Reservoirs

Chamber Airlock

High Voltage **Power Supply**



Wind/Unwind Stage

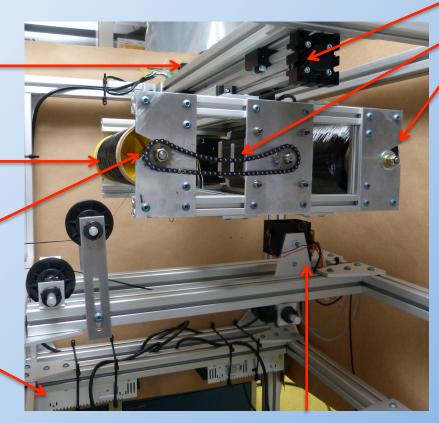
NASA Aeronautics Research Institute

Stepper Motor
Driver, Wind Motor
Driver not visible

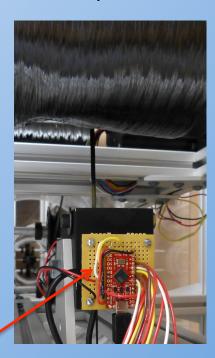
Wind Spool

Quick Change Spool Holders

Stepper Motor Power Supplies



Lateral Stage
Wind Motor
Unwind Spool



Microcontroller and Optical Sensors



Deposition Chamber

NASA Aeronautics Research Institute

Vent Pipes

Purge Air Inlet

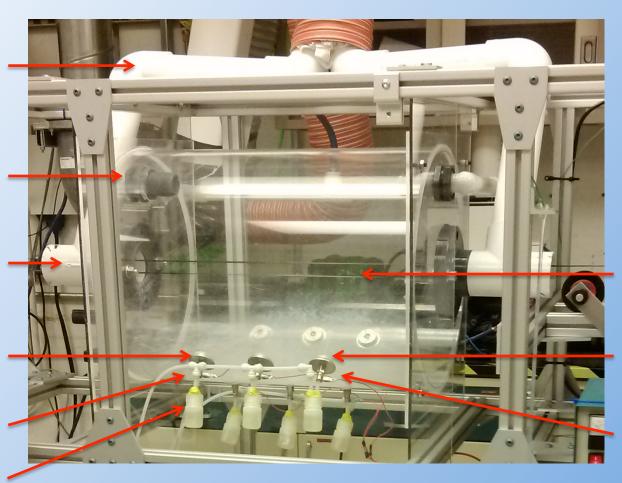
Airlock

Mounting Magnets

Air Tube

Solution

Reservoirs



Carbon Fiber
Towpreg passes
through twice

Needles

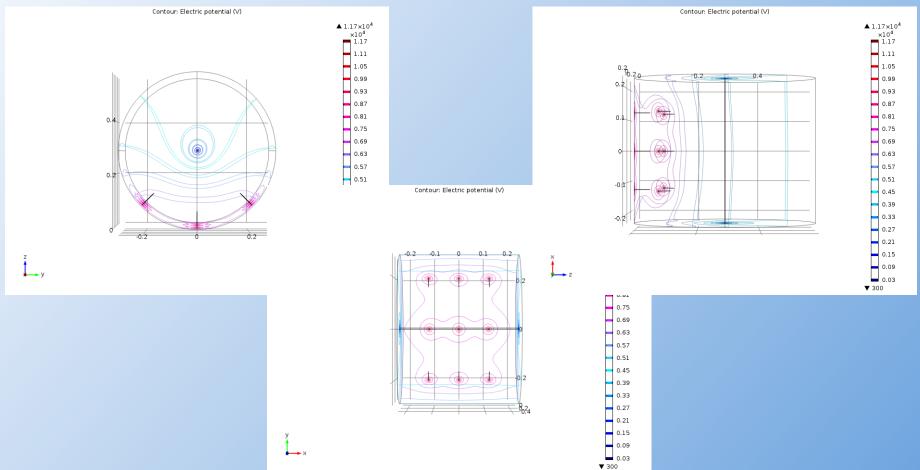
High Voltage Connection



Modeling

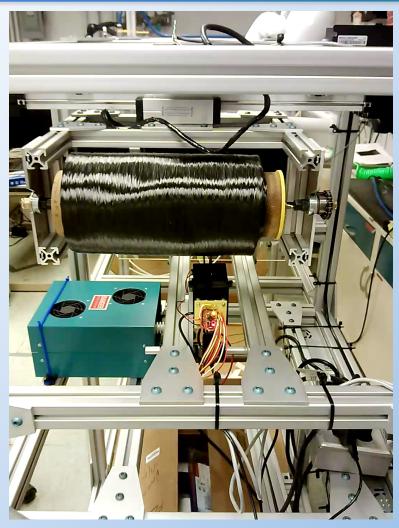
NASA Aeronautics Research Institute

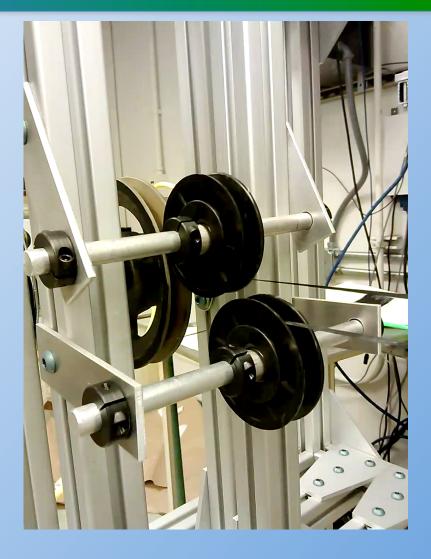
FEA Modeling of Electric Fields with the Selected Needle Array





Nanofiber Deposition



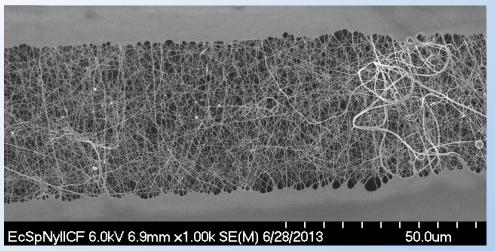




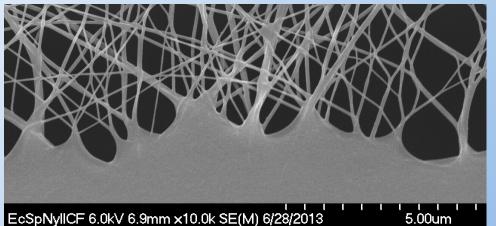
Images of Coated Tow

NASA Aeronautics Research Institute

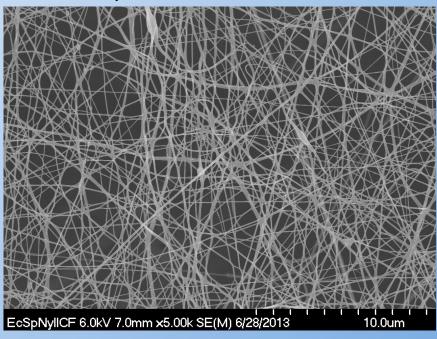
Nanofiber on towpreg spanning a valley



Wetting of the nanofiber by the epoxy resin



Nylon 11 nanofiber mesh

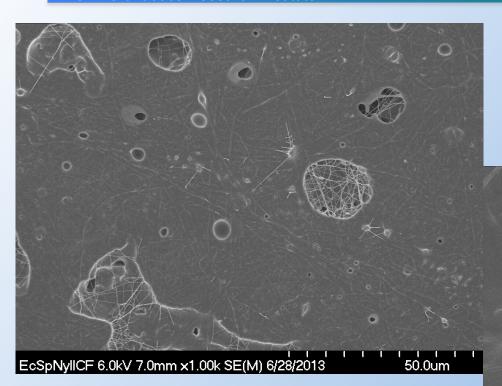


Most of the nanofibers are between 100 and 300 nanometers. The fibers are very interconnected indicating that solvent was still present when they came in contact. Further process improvement may be needed.



Images of Coated Tow

NASA Aeronautics Research Institute



Nylon 11 nanofiber mesh on the

towpreg surface that has been pressed

into the epoxy resin by the rollers

Images were acquired by Scanning Electron Microscopy (SEM) at GRC.

EcSpNyllCF 6.0kV 6.9mm ×10.0k SE(M) 6/28/2013 5.00um



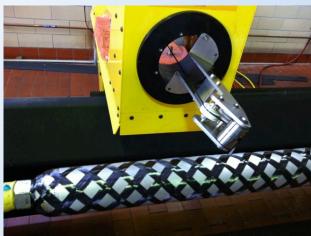
+/-45 Degree

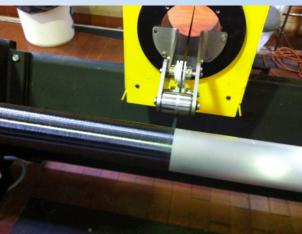
90 Degree

Fabrication of Test Coupons

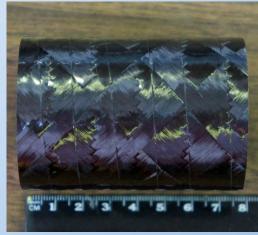
NASA Aeronautics Research Institute

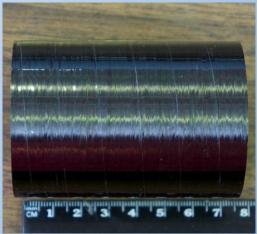
Filament Winding





Compression Specimens





Tension Specimens





Mechanical Testing

Architecture	Baseline 90	Modified 90	Baseline +/-45	Modified +/-45
Tension	5	5	5	5
Compression	5	5	5	5
Post-Impact Compression	5	5	5	5

- +/-45 tension and compression will address Mode 3 shear of the interface.
- 90 tension will address transverse tensile strength.
- Post-impact compression will address resistance to damage propagation and distribution of damage due to impact loads through the thickness.



Tension Fixture



Accomplishments

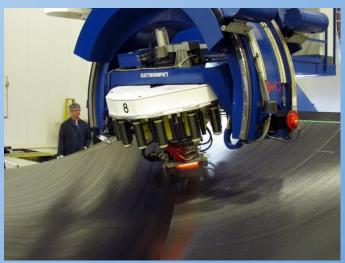
- Scaled up the deposition method to a continuous process.
- Successfully electrospun Nylon 11 onto carbon fiber towpreg.
- Produced 1000's of feet of nanofiber coated towpreg.
- Had filament wound tubes produced with baseline and coated material.
- Submitted a full patent application on June 14, 2013 titled "System and Method for Coating A Tow With An Electrospun Nanofiber."
- Began mechanical testing which will be completed before the end of July, 2013.



Path Forward

- Additional work has been proposed that would focus on the selection and optimization of the nanofiber material and fracture toughness testing.
- Other nano-additives could be used in the precursor solution such as nanoparticles and nanotubes to provide a stable means of deposition while isolating the nano-material in the nanofiber.
- Wider continuous material could also be coated using this method.
- Properties other than toughness could be modified with this approach, including possibly thermal conductivity.







Thank you.

Presented by Dr. Lee W. Kohlman NASA Glenn Research Center